Lattice Boltzmann model for axisymmetric thermal flows

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A thermal lattice Boltzmann (LB) model is presented for axisymmetric thermal flows in the incompressible limit. The model is based on the double-distribution-function LB method, which has attracted much attention since its emergence for its excellent numerical stability over the multispeed LB method. Compared with the existing axisymmetric thermal LB models, the present model is simpler and retains the inherent features of the standard LB method. Numerical simulations are carried out for the thermally developing laminar flows in circular ducts and the natural convection in an annulus between two coaxial vertical cylinders. The Nusselt number obtained from the simulations agrees well with the analytical solutions and/or the results reported in previous studies.

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In recent years, the lattice Boltzmann (LB) method for simulating axisymmetric flows has attracted much attention [1-10]. In fact, the LB simulation of axisymmetric flows can be directly handled with a three-dimensional (3D) LB model. However, such a treatment does not take the advantage of the axisymmetric property of the flow: a 3D axisymmetric flow can be reduced to a quasi-two-dimensional (2D) problem. To make use of this feature, Halliday et al. [1] first studied the 2D LB method for axisymmetric flows in 2001. Some source terms containing density and velocity gradients were introduced into the microscopic evolution equation. However, this method fails to reproduce the correct hydrodynamic momentum equation due to some missing terms: the term $\rho u_i u_r / r$ is missing in the recovered momentum equation and some additional terms involving the first-order source term are missing in the second-order expansion of the microscopic evolution equation. These missing terms were noticed by Lee et al. [2] and Reis *et al.* [3,4]. By adding these terms, Lee *et al.* developed a more accurate axisymmetric LB model. Reis et al. rederived Halliday et al.'s model and then presented a modified version. Zhou [5] recently proposed a simplified axisymmetric isothermal model, in which the source terms are simple, yet still contain a velocity gradient term which should be determined with a finite-difference scheme. Most recently, Guo et al. [6] developed a simple and consistent LB model for axisymmetric isothermal flows based on the continuous Boltzmann equation. The source term in the model contains no gradients and is easier to implement.

There are also several attempts for constructing axisymmetric thermal LB models. The first attempt was made by Peng *et al.* [7] through the hybrid LB approach. In their model, the azimuthal velocity and the temperature field are solved by the second-order center difference scheme. Later, Huang *et al.* [8] found that, for flows with high Reynolds number and Rayleigh number, the convection terms in the Navier-Stokes equations become dominant and the second-order center difference scheme is unsuitable due to the enhanced numerical instability. Then they proposed an im-

In the literature, Lallemand and Luo [11] have pointed out that the hybrid LB approach significantly deviates from the standard LB method, which means it loses some inherent features of the standard LB method, and it only provides a compromised solution. Alternatively, the double-distributionfunction (DDF) LB approach [12-17], which utilizes two different distribution functions, one for the velocity field and the other for the temperature or energy field, has attracted much attention since its emergence for its excellent numerical stability as well as the retaining of inherent features of the standard LB method. The aim of this study is to develop a thermal LB model for simulating axisymmetric thermal flows based on the DDF LB approach. The velocity field of the incompressible axisymmetric thermal flows can be solved with isothermal axisymmetric LB models. In what follows we focus on discussing the microscopic evolution equation for solving the temperature field.

The macroscopic temperature equation of incompressible axisymmetric thermal flows in a cylindrical coordinate system can be written as

$$\partial_t T + u_i \partial_i T = \partial_i (\chi \partial_i T) + \chi \frac{1}{r} \partial_r T, \qquad (1)$$

where *T* is the temperature; *i* indicates the *r* or *x* component, here *r* and *x* are the coordinates in radial and axial directions, respectively; u_i is the component of velocity in the *i* direction.

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proved version of Peng *et al.*'s model. Recently, Chen *et al.* [9,10] pointed out that, although Huang *et al.*'s hybrid LB model is more numerically stable than Peng *et al.*'s model, too many complicated source terms exist in their model and a great deal of lattice grids are still required for numerical stability. Noticing this problem, they devised a thermal LB model for axisymmetric thermal flows based on the vorticity-stream-function (VSF) equations [10]. The source terms are simplified by invoking the VSF formulation but still contain several gradient terms. Guo *et al.* recently argued that Chen *et al.*'s model will become very inefficient for unsteady flows because a Poisson equation must be solved at every time step [6]. Meanwhile, the boundary condition is not easy to implement in the VSF-based numerical methods.

tion; and χ is the thermal diffusivity. With the continuity equation $\partial_i u_i = -u_r/r$, we can rewritten Eq. (1) as

$$\partial_t T + \partial_i (u_i T) = \partial_i (\chi \partial_i T) + \chi \frac{1}{r} \partial_r T - \frac{u_r T}{r}.$$
 (2)

The underlined terms arise from the cylindrical polar coordinates. In order to recover these terms, we introduce the following temperature evolution equation:

$$g_{\alpha}(\mathbf{r} + \mathbf{e}_{\alpha}\delta_{t}, t + \delta_{t}) - g_{\alpha}(\mathbf{r}, t)$$

$$= -\frac{1}{2\tau_{g}} [g_{\alpha}(\mathbf{r} + \mathbf{e}_{\alpha}\delta_{t}, t + \delta_{t}) - g_{\alpha}^{eq}(\mathbf{r} + \mathbf{e}_{\alpha}\delta_{t}, t + \delta_{t})]$$

$$-\frac{1}{2\tau_{g}} [g_{\alpha}(\mathbf{r}, t) - g_{\alpha}^{eq}(\mathbf{r}, t)] - \frac{e_{\alpha r}}{r} \delta_{t} [g_{\alpha}(\mathbf{r}, t) - g_{\alpha}^{eq}(\mathbf{r}, t)]$$

$$+ \frac{\delta_{t}}{2} [S_{\alpha}(\mathbf{r} + \mathbf{e}_{\alpha}\delta_{t}, t + \delta_{t}) + S_{\alpha}(\mathbf{r}, t)], \qquad (3)$$

where g_{α} is the temperature distribution function; τ_g is nondimensional relaxation time for the temperature field; S_{α} is the source term; and the discrete velocities { e_{α} = $(e_{\alpha x}, e_{\alpha r})$: α =0,1,...,8} are specified by the standard D2Q9 lattice. The underlined term in Eq. (3) is used to recover the second term on the right-hand side of Eq. (2). Actually, we can prove that, when a similar treatment is combined with Zhou's isothermal axisymmetric model [5], the source terms of the rearranged model will contain no gradient terms.

 g_{α}^{eq} is chosen as $g_{\alpha}^{eq} = T f_{\alpha}^{eq} = \rho T w_{\alpha} [1 + (e_{\alpha} \cdot u)/c_s^2 + 0.5(e_{\alpha} \cdot u)^2/c_s^4 - 0.5u^2/c_s^2]$, where $c_s = c/\sqrt{3}$ $(c = \delta_x/\delta_t$ is the lattice speed) is the sound speed and the weights w_{α} are given by $w_0 = 4/9$, $w_{1-4} = 1/9$, and $w_{5-8} = 1/36$. It can be found that g_{α}^{eq} satisfies

$$\sum_{\alpha} g_{\alpha}^{eq} = \rho T, \quad \sum_{\alpha} e_{\alpha i} g_{\alpha}^{eq} = \rho T u_i, \tag{4}$$

$$\sum_{\alpha} e_{\alpha i} e_{\alpha j} g_{\alpha}^{eq} = \rho T u_i u_j + p T \delta_{ij}.$$
 (5)

Through the second-order Taylor-series expansion, evolution Eq. (3) can be reduced to

$$\begin{split} \delta_t(\partial_t + \boldsymbol{e}_{\alpha} \cdot \boldsymbol{\nabla}) g_{\alpha} + \frac{\delta_t^2}{2} (\partial_t + \boldsymbol{e}_{\alpha} \cdot \boldsymbol{\nabla})^2 g_{\alpha} \\ &= -\frac{1}{\tau_g} (g_{\alpha} - g_{\alpha}^{eq}) - \frac{\delta_t}{2\tau_g} (\partial_t + \boldsymbol{e}_{\alpha} \cdot \boldsymbol{\nabla}) (g_{\alpha} - g_{\alpha}^{eq}) \\ &- \frac{e_{\alpha r}}{r} \delta_t (g_{\alpha} - g_{\alpha}^{eq}) + \delta_t S_{\alpha} + \frac{\delta_t^2}{2} (\partial_t + \boldsymbol{e}_{\alpha} \cdot \boldsymbol{\nabla}) S_{\alpha} + O(\delta_t^3), \end{split}$$

$$(6)$$

where $\nabla = (\partial_x, \partial_r)$ is the spatial gradient operator. By introducing the following expansions [18]:

$$\partial_t = \partial_{t0} + \delta_t \partial_{t1}, \quad g_\alpha = g_\alpha^{(0)} + \delta_t g_\alpha^{(1)} + \delta_t^2 g_\alpha^{(2)}, \tag{7}$$

we can rewrite Eq. (6) in the consecutive orders of δ_t as

$$O(1):g_{\alpha}^{(0)} = g_{\alpha}^{eq}, \tag{8}$$

$$O(\delta_t): (\partial_{t0} + \boldsymbol{e}_{\alpha} \cdot \nabla) g_{\alpha}^{(0)} + \frac{1}{\tau_g} g_{\alpha}^{(1)} = S_{\alpha}, \qquad (9)$$

$$O(\delta_{t}^{2}):\partial_{t1}g_{\alpha}^{(0)} + (\partial_{t0} + \boldsymbol{e}_{\alpha} \cdot \nabla)g_{\alpha}^{(1)} + \frac{1}{2}(\partial_{t0} + \boldsymbol{e}_{\alpha} \cdot \nabla)^{2}g_{\alpha}^{(0)} + \frac{1}{\tau_{g}}g_{\alpha}^{(2)} + \frac{1}{2\tau_{g}}(\partial_{t0} + \boldsymbol{e}_{\alpha} \cdot \nabla)g_{\alpha}^{(1)} = -\frac{\boldsymbol{e}_{\alpha r}}{r}g_{\alpha}^{(1)} + \frac{1}{2}(\partial_{t0} + \boldsymbol{e}_{\alpha} \cdot \nabla)S_{\alpha},$$
(10)

Using Eq. (9), we can rewrite Eq. (10) as

$$\partial_{t1}g_{\alpha}^{(0)} + (\partial_{t0} + \boldsymbol{e}_{\alpha} \cdot \nabla)g_{\alpha}^{(1)} + \frac{1}{\tau_g}g_{\alpha}^{(2)} = -\frac{\boldsymbol{e}_{\alpha r}}{r}g_{\alpha}^{(1)}.$$
 (11)

Taking the summations of Eqs. (9) and (11), we can obtain, respectively,

$$\partial_{t0}(\rho T) + \partial_j(\rho u_j T) = \sum_{\alpha} S_{\alpha}, \qquad (12)$$

$$\partial_{t1}(\rho T) + \partial_i \left(\sum_{\alpha} e_{\alpha i} g_{\alpha}^{(1)}\right) = -\frac{1}{r} \sum_{\alpha} e_{\alpha r} g_{\alpha}^{(1)}.$$
 (13)

To recover the target macroscopic temperature equation, $\Sigma_{\alpha}S_{\alpha}$ should be given by $\Sigma_{\alpha}S_{\alpha} = -\rho T u_r/r$. From Eq. (9), we have

$$\sum_{\alpha} e_{\alpha i} g_{\alpha}^{(1)} = -\tau_g \left(\partial_{t_0} \sum_{\alpha} e_{\alpha i} g_{\alpha}^{(0)} + \partial_j \sum_{\alpha} e_{\alpha i} e_{\alpha j} g_{\alpha}^{(0)} \right) + \tau_g \sum_{\alpha} e_{\alpha i} S_{\alpha}.$$
(14)

From Eqs. (4), (5), and (8), we can obtain

$$\partial_{t0} \left(\sum_{\alpha} e_{\alpha i} g_{\alpha}^{(0)} \right) = u_i \partial_{t0} (\rho T) + \rho T \partial_{t0} u_i, \tag{15}$$

$$\partial_j \left(\sum_{\alpha} e_{\alpha i} e_{\alpha j} g_{\alpha}^{(0)} \right) = u_i \partial_j (\rho T u_j) + \rho T u_j \partial_j u_i + T \partial_i p + p \partial_i T,$$
(16)

where $\partial_{t0}u_i$ is evaluated as

$$\partial_{t0}u_i = \left[\partial_{t0}(\rho u_i) - u_i\partial_{t0}\rho\right]/\rho = -u_j\partial_j u_i - (\partial_i p)/\rho.$$
(17)

According to Eqs. (15)–(17), we can rewrite Eq. (14) as

$$\sum_{\alpha} e_{\alpha i} g_{\alpha}^{(1)} = -\tau_g \bigg[u_i \partial_{i0}(\rho T) + u_i \partial_j (\rho T u_j) - \sum_{\alpha} e_{\alpha i} S_{\alpha} + p \partial_i T \bigg].$$
(18)

If we carefully choose $\sum_{\alpha} e_{\alpha i} S_{\alpha} = -\rho T u_i u_r r$, then Eq. (18) can be reduced to $\sum_{\alpha} e_{\alpha i} g_{\alpha}^{(1)} = -\tau_g p \partial_i T$. Therefore in this study the source term S_{α} is chosen as

$$S_{\alpha} = -\frac{u_r}{r} g_{\alpha}^{eq}.$$
 (19)

In the presence of a body force $(F = \rho a)$, actually a forcing term $F_{\alpha} = \rho T w_{\alpha} (e_{\alpha} \cdot a) / c_s^2$ should also be considered in Eq.



FIG. 1. Local Nusselt number distribution along the axial direction for the thermally developing flow.

(3). But this term seemingly can be neglected in most cases [12–14]. Substituting the equation $\sum_{\alpha} e_{\alpha i} g_{\alpha}^{(1)} = -\tau_g p \partial_i T$ into Eq. (13) and then combining Eq. (12) with Eq. (13) $(\partial_t = \partial_{t0} + \delta_t \partial_{t1})$, we can obtain the following macroscopic temperature equation:

$$\partial_t(\rho T) + \partial_i(\rho u_i T) = \partial_i(\rho \chi \partial_i T) + \rho \chi \frac{1}{r} \partial_r T - \frac{\rho u_r T}{r}, \quad (20)$$

where the thermal diffusivity χ is given by $\chi = \delta_t \tau_g c^2/3$. In the incompressible limit with $\rho \approx \rho_0$, Eq. (20) is just the target macroscopic temperature equation. To this end, we can simply modify g_{α}^{eq} as $g_{\alpha}^{eq} = \rho_0 f_{\alpha}^{eq}/\rho$. For small Mach-number flows, g_{α}^{eq} can be further simplified by neglecting the terms of $O(u^2)$ [14]. In this situation, g_{α}^{eq} based on a D2Q4 lattice with four directions e_1 , e_2 , e_3 , and e_4 can also be used: g_{α}^{eq} $= (\rho_0 T/4)[1+2(e_{\alpha} \cdot u)/c^2]$ together with $\chi = \delta_t \tau_g c^2/2$.

To eliminate the implicitness of Eq. (3), following He *et* al. [13], a distribution function $\overline{g_{\alpha}} = g_{\alpha} + 0.5(g_{\alpha} - g_{\alpha}^{eq})/\tau_g$ $-0.5\delta_t S_{\alpha}$ can be shown. Through some standard algebra, the evolution equation for $\overline{g_{\alpha}}$ can be obtained

$$\frac{\overline{g}_{\alpha}(\mathbf{x} + \boldsymbol{e}_{\alpha}\delta_{t}, t + \delta_{t}) - \overline{g}_{\alpha}(\mathbf{x}, t)}{= -\omega_{g}[\overline{g}_{\alpha}(\mathbf{x}, t) - g_{\alpha}^{eq}(\mathbf{x}, t)] + (1 - 0.5\omega_{g})\delta_{t}S_{\alpha}(\mathbf{x}, t),$$
(21)

where ω_g is given by $\omega_g = [1 + (e_{\alpha r} \tau_g \delta_t / r)] / (\tau_g + 0.5)$. The macroscopic temperature can be calculated from the new distribution function as $T = \sum_{\alpha \overline{\delta_{\alpha}}} / [\rho_0(1 + 0.5 \delta_t u_r / r)]$. The velocity field is solved by using isothermal axisymmetric LB models. Here Guo *et al.*'s isothermal axisymmetric LB model is adopted, which can be briefly summarized as follows [6]: the evolution equation for velocity field is

$$f_{\alpha}(\mathbf{x} + \mathbf{e}_{\alpha}\delta_{t}, t + \delta_{t}) - f_{\alpha}(\mathbf{x}, t)$$

= $-\omega_{f}[\tilde{f}_{\alpha}(\mathbf{x}, t) - \tilde{f}_{\alpha}^{eq}(\mathbf{x}, t)] + \delta_{t}(1 - 0.5\omega_{f})G_{\alpha}(\mathbf{x}, t),$
(22)

with $\omega_f = 1/(\tau_f + 0.5)$, $G_\alpha = (e_\alpha - u) \cdot \tilde{a} \tilde{f}_\alpha^{eq} / c_s^2$, $\tilde{a}_x = a_x$, $\tilde{a}_r = a_r + c_s^2 [1 - 2\delta_t \tau_f u_r / r] / r$, and $\tilde{f}_\alpha^{eq} = r f_\alpha^{eq}$, where τ_f is the nondimensional relaxation time for the velocity field; a_x and a_r are the components of the external force acceleration in the x and r directions, respectively. The macroscopic density and



FIG. 2. Natural convection between coaxial vertical cylinders.

velocity are calculated by $\rho = \sum_{\alpha} \tilde{f}_{\alpha}/r$ and $u_i = r[\sum_{\alpha} e_{\alpha i} \tilde{f}_{\alpha} + 0.5 \delta_t r \rho a_i + 0.5 \delta_t \rho c_s^2 \delta_{ir}]/[\rho(r^2 + \tau_f \delta_i c_s^2 \delta_{ir})]$. The kinematic viscosity is given by $\nu = \delta_t \tau_f c^2/3$. Equations (21) and (22) together with the corresponding equilibrium distributions and the source terms constitute the present thermal DDF LB model for axisymmetric thermal flows.

Two numerical tests are considered to validate the proposed model. The first test is the thermally developing flow in a circular duct. A uniform temperature profile $T_{in} = 10$ and a thermally fully developed flow are respectively imposed at the inlet and outlet. Two different thermal boundary conditions (BCs), the constant wall temperature BC (type 1) and the constant wall heat flux BC (type 2), are considered at the wall. In simulations, the relaxation time $\tau_f = 0.6$, the Prandtl number $Pr = \nu / \chi = 0.7$, and a grid size of $N_r \times N_r = 649 \times 82$ is adopted, corresponding to an aspect ratio L/D=648/81=8. The periodic BC is applied in the axial direction for the velocity BCs with a body force $\rho a_r = \rho \times 10^{-4}$, while the nonequilibrium extrapolation BC is applied at the inlet and the wall for the thermal BCs. Meanwhile, the outflow is supposed to be fully developed and to obey the Neumann rule. The contours of the local Nusselt number, which is defined as $\operatorname{Nu}_x = -D(\partial_r T)_w / (T_w - T_b)$, where D is the diameter, $T_w = 1$, and $T_b = \int_0^{D/2} 2\pi r u T dr / \int_0^{D/2} 2\pi r u dr$ is the bulk temperature, are plotted in Fig. 1 along the axial direction. The Nusselt numbers are 3.674 and 4.376 respectively in the thermal fully developed region for the two different BCs. Compared with the corresponding analytical solutions [19], 3.66 and 4.36, the relative errors are 0.38% and 0.37%, respectively. Numerical simulations have also been conducted with the scheme described in Ref. [7]. The obtained Nusselt numbers are 3.710 and 4.418, respectively. The corresponding relative errors are 1.37% and 1.33%. Meanwhile, the time step of the



FIG. 3. Streamlines for $Ra=10^4$ (a) and 10^5 (b).



FIG. 4. Isotherms for $Ra=10^4$ (a) and 10^5 (b).

finite-difference scheme is set to be $0.5 \delta_t$ due to the numerical instability with δ_t . The comparison shows that the present scheme is more accurate and stable.

The second test is the natural convection in an annulus between two coaxial vertical cylinders [20,21]. The problem is sketched in Fig. 2, where g is the gravitation acceleration, T_i and T_o are the constant temperatures of the inner and outer cylinders, respectively, and $T_i > T_o$. The radius ratio r_o / r_i and the aspect ratio $h/(r_o - r_i)$ are both set to be 2.0. The natural convection is characterized by the Prandtl number $Pr = \nu / \chi$ and the Rayleigh number $\operatorname{Ra} = g\beta(T_i - T_o)(r_o - r_i)^3 \operatorname{Pr}/\nu^2$, where β is the thermal-expansion coefficient. The buoyancy force is given by $\rho a_r = -\rho g(T - T_r)$, where $T_r = (T_i + T_o)/2$. Numerical simulations are carried out for $Ra=10^4$ and 10^5 . A grid size of $N_r \times N_r = 101 \times 201$ is adopted. The streamlines and isotherms at the steady state are shown in Fig. 3 and 4, respectively. To quantify the results, in Table I, the Nusselt numbers defined as Nu_{*i*,*o*} = $-(1/h)r_{i,o}\int_{0}^{h} (\partial_{x}T)_{i,o}/(T_{i}-\underline{T_{o}})dx$ are compared with the average Nusselt number (Nu=(Nu)) $+Nu_o)/2$ reported in Refs. [20,21]. Meanwhile, the average Nusselt numbers obtained from the D2Q4 lattice are 3.219

TABLE I. Comparison of the Nusselt number.

Ra	Ref. [20]	Ref. [21]	Nu _i	Nu _o
10 ⁴	3.037	3.163	3.216	3.218
10 ⁵	5.760	5.882	5.782	5.787

and 5.782 respectively for Ra=10⁴ and 10⁵, which are in good agreement with the results obtained from the D2Q9 lattice with $g_{\alpha}^{eq} = \rho_0 f_{\alpha}^{eq} / \rho$.

In summary, a thermal LB model has been presented for simulating laminar axisymmetric thermal flows of incompressible fluids with negligible viscous heat dissipation. The source terms of the model contain no gradient terms. Compared with existing models, the present model is simpler and retains the inherent features of the standard LB method. Numerical experiments show that axisymmetric thermal flows can be well simulated. In addition, it has been found that the present approach can be extended to solve the azimuthal velocity field of axisymmetric rotating or swirling flows. We found that the macroscopic equation of azimuthal velocity u_{θ} can be rewritten as $\partial_t U_{\theta} + \partial_i (u_i U_{\theta}) = \partial_i (\nu \partial_i U_{\theta}) - 3\nu \partial_r U_{\theta} / r$ $+3\nu U_{\theta}/r^2$ with $U_{\theta}=r^2u_{\theta}$ [22]. Here it can be clearly seen that the equation of U_{θ} is very similar to Eq. (2). Then a similar simple LB evolution equation (D2O9 and D2O4) without gradient terms can be easily obtained according to the formulations in the present Brief Report [23].

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